
TEMA: 0115	ATP - (CHAP. 04) PERFORMANCE	
COD_PREG: 8117	PREGUNTA: Fig. 1 What is the maximum landing distance that may be used by a turbopropeller-powered, small transport category airplane to land on Rwy 24 (Dry) at the alternate airport?	RPTA: C
OPCION A: 5,490 feet		
OPCION B: 6,210 feet		
OPCION C: 6,405 feet		
(Ver figura referencial 1 en el Manual de Figuras)		
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8118	Fig. 1 What is the maximum landing distance that may be used by a reciprocating-engine-powered, small transport category airplane to land on Rwy 24 (dry) at the destination airport?	A
OPCION A: 5,490 feet.		
OPCION B: 6,210 feet.		
OPCION C: 6,405 feet.		
(Ver figura referencial 1 en el Manual de Figuras)		
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8119	Fig. 1 What is the maximum landing distance that may be used by a turbopropeller-powered small transport category airplane to land on Rwy 6 (dry) at the alternate airport?	C
OPCION A: 5,460 feet		
OPCION B: 6,210 feet		
OPCION C: 6,370 feet		
(Ver figura referencial 1 en el Manual de Figuras)		
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8120	Fig. 1 What is the maximum landing distance that may be used by a reciprocating-engine-powered, small transport category airplane to land on Rwy 6 (dry) at the destination airport?	A
OPCION A: 5,460 feet		
OPCION B: 6,210 feet		
OPCION C: 6,370 feet		
(Ver figura referencial 1 en el Manual de Figuras)		
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8121	Fig. 1 What is the maximum landing distance that may be used by a turbine-engine-powered, small transport category airplane to land on Rwy 24 (dry) at the destination airport?	B
OPCION A: 5,460 feet		
OPCION B: 5,490 feet		
OPCION C: 6,210 feet		
(Ver figura referencial 1 en el Manual de Figuras)		
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8122	Fig. 1 What is the maximum landing distance that may be used by a turbine-engine-powered, small transport category airplane to land on Rwy 6 (wet) at the destination airport?	A
OPCION A: 5,460 feet.		
OPCION B: 5,880 feet.		
OPCION C: 6,088 feet.		
(Ver figura referencial 1 en el Manual de Figuras)		
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8123	Fig. 2 What is the maximum landing distance that may be used by a turbopropeller-powered, small transport category airplane to land on Rwy 19 (dry) at the destination airport?	C
OPCION A: 6,020 feet.		
OPCION B: 5,820 feet.		
OPCION C: 5,160 feet.		
(Ver figura referencial 2 en el Manual de Figuras)		

8124 Fig. 2

A

What is the maximum landing distance that may be used by a reciprocating-engine-powered, small transport category airplane to land on Rwy 1 (dry) at the destination airport?

OPCION A: 5,010 feet.

OPCION B: 5,820 feet.

OPCION C: 5,845 feet.

(Ver figura referencial 2 en el Manual de Figuras)

8125 Fig. 2

A

What is the maximum landing distance that may be used by a turbine-engine-powered, small transport category airplane to land on Rwy 1 (dry) at the destination airport?

OPCION A: 5,010 feet.

OPCION B: 5,820 feet.

OPCION C: 5,845 feet.

(Ver figura referencial 2 en el Manual de Figuras)

8126 Fig. 2

A

What is the maximum landing distance that may be used by a turbine-engine-powered, small transport category airplane to land on Rwy 19 (dry) at the destination airport?

OPCION A: 5,160 feet.

OPCION B: 5,820 feet.

OPCION C: 6,020 feet.

(Ver figura referencial 2 en el Manual de Figuras)

8127 Fig. 2

A

May a small transport category, turbine-engine-powered airplane that has a computed landing distance of 5,500 feet use one or both of the runways depicted in the illustration at the destination airport?

OPCION A: Neither Rwy 1 nor Rwy 19 may be used if dry conditions exist.

OPCION B: Only Rwy 19 may be used provided dry conditions exist.

OPCION C: Rwy 1 or Rwy 19 may be used whether conditions are wet or dry.

(Ver figura referencial 2 en el Manual de Figuras)

8128 Fig. 2

B

May a small transport category, turboprop airplane that has a computed landing distance of 6,000 feet use either or both runways depicted in the illustration at the destination airport?

OPCION A: Only Rwy 19 may be used if dry conditions exist.

OPCION B: Neither Rwy 1 nor Rwy 19 may be used under any conditions.

OPCION C: Either Rwy 1 or Rwy 19 may be used whether conditions are wet or dry.

(Ver figura referencial 2 en el Manual de Figuras)

8129 Fig. 2

B

What is the maximum landing distance that may be used for a non-transport category, turbopropeller-driven airplane to land on Rwy 1 (dry) at the alternate airport?

OPCION A: 5,010 feet.

OPCION B: 5,845 feet.

OPCION C: 6,020 feet.

(Ver figura referencial 2 en el Manual de Figuras)

8130 Fig.2

A

Which condition meets FAR Part 135 operational requirements for a small, transport category, turboprop airplane to land at the destination airport that has the runway environment given in the illustration?

OPCION A: The airport may be listed as the destination airport if the landing distance does not exceed 5,160 feet for Rwy 19.

OPCION B: The airport may NOT be listed as the destination airport if the landing distance exceeds 5,100 feet for Rwy 19.

OPCION C: The airport may be listed as the destination airport if the landing distance does not exceed 5,350 feet for either runway, wet or dry conditions.

(Ver figura referencial 2 en el Manual de Figuras)

8133	What effective runway length is required for a turbojet-powered airplane at the destination airport if the runways are forecast to be wet or slippery at the ETA?	B
OPCION A:	70 percent of the actual runway available, from a height of 50 feet over the threshold.	
OPCION B:	115 percent of the runway length required for a dry runway.	
OPCION C:	115 percent of the runway length required for a wet runway.	

8134	For which of these aircraft is the "clearway" for a particular runway considered in computing takeoff weight limitations?	B
OPCION A:	Those passenger-carrying transport aircraft certificated between August 26, 1957 and August 30, 1959.	
OPCION B:	Turbine-engine-powered transport airplanes certificated after September 30, 1958.	
OPCION C:	U.S. certified air carrier airplanes certificated after August 29, 1959.	

8344	How can turbulent air cause an increase in stalling speed of an airfoil?	A
OPCION A:	An abrupt change in relative wind.	
OPCION B:	A decrease in angle of attack.	
OPCION C:	Sudden decrease in load factor.	

8369	If an engine failure occurs at an altitude above single-engine ceiling, what airspeed should be maintained?	B
OPCION A:	Vmc.	
OPCION B:	Vyse.	
OPCION C:	Vxse.	

8370	What is the resulting performance loss when one engine on a twin-engine airplane fails?	B
OPCION A:	Reduction of cruise airspeed by 50 percent.	
OPCION B:	Reduction of climb by 50 percent or more.	
OPCION C:	Reduction of all performance by 50 percent.	

8371	Under what condition is Vmc the highest?	B
OPCION A:	Gross weight is at the maximum allowable value.	
OPCION B:	CG is at the most rearward allowable position.	
OPCION C:	CG is at the most forward allowable position.	

8374	What effect does landing at high elevation airports have on groundspeed with comparable conditions relative to temperature, wind, and airplane weight?	A
OPCION A:	Higher than at low elevation.	
OPCION B:	Lower than at low elevation.	
OPCION C:	The same as at low elevation.	

8381	Which maximum range factor decreases as weight decreases?	C
OPCION A:	Angle of attack.	
OPCION B:	Altitude.	
OPCION C:	Airspeed.	

8383	What performance is characteristic of flight at maximum L/D in a propeller-driven airplane?	A
OPCION A:	Maximum range and distance glide.	
OPCION B:	Best angle of climb.	
OPCION C:	Maximum endurance.	

8398	What should a pilot do to maintain "best range" airplane performance when a tailwind is encountered?	C
OPCION A:	Increase speed.	
OPCION B:	Maintain speed.	
OPCION C:	Decrease speed.	

8400	At what speed, with reference to L/Dmax, does maximum rate-of-climb for a jet airplane occur?	A
OPCION A:	A speed greater than that for L/Dmax.	
OPCION B:	A speed equal to that for L/Dmax.	
OPCION C:	A speed less than that for L/Dmax.	

8401	At what speed, with reference to L/Dmax, does maximum range for a jet airplane occur?	C
OPCION A:	A speed less than that for L/Dmax.	

- OPCION B:** A speed equal to that for L/Dmax.
OPCION C: A speed greater than that for L/Dmax.

8459 Fig. 12 A
Given the following conditions, what is the minimum torque for takeoff?

Pressure altitude 9,000 ft
Tempertaure (OAT) +3°C
Ice vanes Extended

- OPCION A:** 3,100 foot-pound.
OPCION B: 3,040 foot-pound.
OPCION C: 3,180 foot-pound.

(Ver figura referencial 12 en el Manual de Figuras)

8460 Fig. 12 A
Given the following conditions, what is the minimum torque for takeoff?

Pressure altitude 7,500 ft
Tempertaure (OAT) +35°C
Ice vanes Retracted

- OPCION A:** 2,820 foot-pound.
OPCION B: 2,880 foot-pound.
OPCION C: 2,780 foot-pound.

(Ver figura referencial 12 en el Manual de Figuras)

8461 Fig. 12 B
Given the following conditions, what is the minimum torque for takeoff?

Pressure altitude 7,500 ft
Tempertaure (OAT) +9°C
Ice vanes Extended

- OPCION A:** 3,200 foot-pound.
OPCION B: 3,160 foot-pound.
OPCION C: 3,330 foot-pound.

(Ver figura referencial 12 en el Manual de Figuras)

8462 Fig. 12 B
Given the following conditions, what is the minimum torque for takeoff?

Pressure altitude 3,500 ft
Tempertaure (OAT) +43°C
Ice vanes Retracted

- OPCION A:** 3,000 foot-pound.
OPCION B: 3,050 foot-pound.
OPCION C: 3,110 foot-pound.

(Ver figura referencial 12 en el Manual de Figuras)

8463 Fig. 12 C
Given the following conditions, what is the minimum torque for takeoff?

Pressure altitude 5,500 ft
Tempertaure (OAT) +29°C
Ice vanes Retracted

- OPCION A:** 2,950 foot-pound.
OPCION B: 3,100 foot-pound.
OPCION C: 3,200 foot-pound.

(Ver figura referencial 12 en el Manual de Figuras)

8464 Fig. 13

C

Given the following conditions, what is the takeoff distance over a 50-foot obstacle?

Pressure altitude Sea Level
Tempertaure (OAT) +12°C
Weight 16,000 lb
Wind component 16 kts HW
Ice vanes Retracted

OPCION A: 1,750 feet.

OPCION B: 2,800 feet.

OPCION C: 2,550 feet.

(Ver figura referencial 13 en el Manual de Figuras)

8465 Fig. 13

A

Given the following conditions, what is the takeoff ground roll and V1 speed?

Pressure altitude 4,000 ft
Tempertaure (OAT) 0°C
Weight 15,500 lb
Wind component 10 kts TW
Ice vanes Extended

OPCION A: 2,900 feet, 106 knots.

OPCION B: 4,250 feet, 102 knots.

OPCION C: 2,700 feet, 107 knots.

(Ver figura referencial 13 en el Manual de Figuras)

8466 Fig. 13

B

Given the following conditions, what is the takeoff distance over a 50-foot obstacle?

Pressure altitude 2,000 ft
Tempertaure (OAT) +15°C
Weight 16,600 lb
Wind component Calm
Ice vanes Retracted

OPCION A: 3,400 feet.

OPCION B: 3,700 feet.

OPCION C: 4,200 feet.

(Ver figura referencial 13 en el Manual de Figuras)

8467 Fig. 13

A

Given the following conditions, what is the takeoff ground roll and V1 speed?

Pressure altitude 3,000 ft
Tempertaure (OAT) -10°C
Weight 15,000 lb
Wind component 8 kts TW
Ice vanes Extended

OPCION A: 2,200 feet, 105 knots.

OPCION B: 2,000 feet, 113 knots.

OPCION C: 1,900 feet, 103 knots.

(Ver figura referencial 13 en el Manual de Figuras)

8468 Fig. 13

A

Given the following conditions, what is the takeoff distance over a 50 foot obstacle?

Pressure altitude 6,000 ft
Tempertaure (OAT) +35°C
Weight 14,500 lb
Wind component 10 kts HW
Ice vanes Retracted

OPCION A: 4,150 feet.

OPCION B: 4,550 feet.

OPCION C: 2,600 feet.

(Ver figura referencial 13 en el Manual de Figuras)

8469 Fig. 14

C

Given the following conditions, what is the accelerate-stop field length?

Pressure altitude 5,000 ft
Tempertaure (OAT) +20°C
Weight 15,000 lb
Wind component 10 kts HW
Ice vanes Retracted

OPCION A: 6,300 feet.

OPCION B: 4,700 feet.

OPCION C: 4,300 feet.

(Ver figura referencial 14 en el Manual de Figuras)

8470 Fig. 14

C

Given the following conditions, what is the accelerate-stop field length?

Pressure altitude 2,000 ft
Tempertaure (OAT) -15°C
Weight 16,000 lb
Wind component 5 kts HW
Ice vanes Extended

OPCION A: 3,750 feet.

OPCION B: 4,600 feet.

OPCION C: 4,250 feet.

(Ver figura referencial 14 en el Manual de Figuras)

8471 Fig. 14

A

Given the following conditions, what is the accelerate-stop field length?

Pressure altitude 6,000 ft
Tempertaure (OAT) +10°C
Weight 16,600 lb
Wind component 15 kts HW
Ice vanes Retracted

OPCION A: 4,950 feet.

OPCION B: 4,800 feet.

OPCION C: 5,300 feet.

(Ver figura referencial 14 en el Manual de Figuras)

8472 Fig. 14 B

Given the following conditions, what is the accelerate-stop field length?

Pressure altitude 8,000 ft
Tempertaure (OAT) -5°C
Weight 14,000 lb
Wind component 4 kts TW
Ice vanes Extended

OPCION A: 4,500 feet.

OPCION B: 4,800 feet.

OPCION C: 5,300 feet.

(Ver figura referencial 14 en el Manual de Figuras)

8473 Fig. 14 C

Given the following conditions, what is the accelerate-stop field length?

Pressure altitude Sea Level
Tempertaure (OAT) +30°C
Weight 13,500 lb
Wind component 14 kts HW
Ice vanes Retracted

OPCION A: 2,500 feet.

OPCION B: 2,850 feet.

OPCION C: 3,050 feet.

(Ver figura referencial 14 en el Manual de Figuras)

8474 Fig. 15 - 16 - 17 C

What is the two-engine rate of climb after takeoff in climb configuration for Operating Conditions BE-21?

OPCION A: 1,350 ft/min.

OPCION B: 2,450 ft/min.

OPCION C: 2,300 ft/min.

(Ver figura referencial 15/16/17 en el Manual de Figuras)

8475 Fig. 15 - 16 - 17 C

What is the single-engine climb gradient after takeoff in climb configuration for Operating Conditions BE-22?

OPCION A: 6.8 percent gradient.

OPCION B: 7.5 percent gradient.

OPCION C: 5.6 percent gradient.

(Ver figura referencial 15/16/17 en el Manual de Figuras)

8476 Fig. 15 - 16 - 17 B

What is the two-engine rate of climb after takeoff in climb configuration for Operating Conditions BE-23?

OPCION A: 1,500 ft/min.

OPCION B: 2,600 ft/min.

OPCION C: 2,490 ft/min.

(Ver figura referencial 15/16/17 en el Manual de Figuras)

8477 Fig. 15 - 16 - 17 A

What is the two-engine rate of climb after takeoff in climb configuration for Operating Conditions BE-24?

OPCION A: 2,100 ft/min.

OPCION B: 2,400 ft/min.

OPCION C: 1,500 ft/min.

(Ver figura referencial 15/16/17 en el Manual de Figuras)

8478 Fig. 15 - 16 - 17 C

What is the single-engine rate of climb after takeoff in climb configuration for Operating Conditions BE-25?

OPCION A: 385 ft/min.

OPCION B: 780 ft/min.

OPCION C: 665 ft/min.

OPCION C: 665 ft/min.

(Ver figura referencial 15/16/17 en el Manual de Figuras)

8479 Fig.15 - 18

B

What are the time, fuel, and distance from the start of climb to cruise altitude for Operating Conditions BE-21?

OPCION A: 10.0 minutes; 290 pounds; 35 NM.

OPCION B: 10.0 minutes; 165 pounds; 30 NM.

OPCION C: 11.5 minutes; 165 pounds; 30 NM.

(Ver figura referencial 15/18 en el Manual de Figuras)

8480 Fig. 15 - 18

B

What are the time, fuel, and distance from the start of climb to cruise altitude for Operating Conditions BE-22?

OPCION A: 12.0 minutes; 220 pounds; 40 NM.

OPCION B: 11.0 minutes; 185 pounds; 37 NM.

OPCION C: 10.5 minutes; 175 pounds; 32 NM.

(Ver figura referencial 15/18 en el Manual de Figuras)

8481 Fig. 15 - 18

B

What are the time, fuel, and distance from the start of climb to cruise altitude for Operating Conditions BE-23?

OPCION A: 13.0 minutes; 180 pounds; 35 NM.

OPCION B: 14.0 minutes; 210 pounds; 40 NM.

OPCION C: 15.0 minutes; 240 pounds; 46 NM.

(Ver figura referencial 15/18 en el Manual de Figuras)

8482 Fig. 15 - 18

C

What are the time, fuel, and distance from the start of climb to cruise altitude for Operating Conditions BE-24?

OPCION A: 12.0 minutes; 220 pounds; 45 NM.

OPCION B: 9.0 minutes; 185 pounds; 38 NM.

OPCION C: 10.0 minutes; 170 pounds; 30 NM.

(Ver figura referencial 15/18 en el Manual de Figuras)

8483 Fig. 15 - 18

C

What are the time, fuel, and distance from the start of climb to cruise altitude for Operating Conditions BE-25?

OPCION A: 11.5 minutes; 170 pounds; 31 NM.

OPCION B: 8.0 minutes; 270 pounds; 28 NM.

OPCION C: 12.5 minutes; 195 pounds; 38 NM.

(Ver figura referencial 15/18 en el Manual de Figuras)

8484 Fig. 19 - 20

A

At what altitude is the service ceiling with one engine inoperative for Operating Conditions BE-26?

OPCION A: 13,000 feet.

OPCION B: 14,200 feet.

OPCION C: 13,600 feet.

(Ver figura referencial 19/20 en el Manual de Figuras)

8485 Fig. 19 - 20

B

Which statement is true regarding performance with one engine inoperative for Operating Conditions BE-27?

OPCION A: Climb rate at the MEA is more than 50 ft/min.

OPCION B: Service ceiling is below the MEA.

OPCION C: Bleed air OFF improves service ceiling by 3,000 feet.

(Ver figura referencial 19/20 en el Manual de Figuras)

8486 Fig. 19 - 20

C

At what altitude is the service ceiling with one engine inoperative for Operating Conditions BE-28?

OPCION A: 1,500 feet above the MEA.

OPCION B: 10,400 feet.

OPCION C: 11,800 feet.

(Ver figura referencial 19/20 en el Manual de Figuras)

8487 Fig. 19 - 20

A

Which statement is true regarding performance with one engine inoperative for Operating Conditions BE-29?

OPCION A: Service ceiling is more than 100 feet above the MEA.

OPCION B: Bleed air must be OFF to obtain a rate of climb of 50 ft/min at the MEA.

OPCION C: Climb is not possible at the MEA.

(Ver figura referencial 19/20 en el Manual de Figuras)

8488 Fig. 19 - 20

C

At what altitude is the service ceiling with one engine inoperative for Operating Conditions BE-30?

OPCION A: 9,600 feet.

OPCION B: 13,200 feet.

OPCION C: 2,100 feet above the MEA.

(Ver figura referencial 19/20 en el Manual de Figuras)

8489 Fig. 21 to 25

B

What is the en route time of the cruise leg for Operating Conditions BE-31?

OPCION A: 1 hour 11 minutes.

OPCION B: 1 hour 17 minutes.

OPCION C: 1 hour 19 minutes.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8490 Fig. 21 to 25

A

What is the en route time of the cruise leg for Operating Conditions BE-32?

OPCION A: 1 hour 13 minutes.

OPCION B: 1 hour 15 minutes.

OPCION C: 1 hour 20 minutes.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8491 Fig. 21 to 25

C

What is the en route time of the cruise leg for Operating Conditions BE-33?

OPCION A: 1 hour 50 minutes.

OPCION B: 1 hour 36 minutes.

OPCION C: 1 hour 46 minutes.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8492 Fig. 21 to 25

A

What is the en route time of the cruise leg for Operating Conditions BE-34?

OPCION A: 1 hour 6 minutes.

OPCION B: 1 hour 3 minutes.

OPCION C: 1 hour 11 minutes.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8493 Fig. 21 to 25

C

What is the en route time of the cruise leg for Operating Conditions BE-35?

OPCION A: 1 hour 6 minutes.

OPCION B: 1 hour 8 minutes.

OPCION C: 1 hour 10 minutes.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8494 Fig. 21 to 25

A

What is the fuel consumption during the cruise leg for Operating Conditions BE-31?

OPCION A: 812 pounds.

OPCION B: 749 pounds.

OPCION C: 870 pounds.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8495 Fig. 21 to 25

C

What is the fuel consumption during the cruise leg for Operating Conditions BE-32?

OPCION A: 1,028 pounds.

OPCION B: 896 pounds.

OPCION C: 977 pounds.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8496 Fig. 21 to 25

B

What is the fuel consumption during the cruise leg for Operating Conditions BE-33?

OPCION A: 1,165 pounds.

OPCION B: 1,373 pounds.

OPCION C: 976 pounds.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8497 Fig. 21 to 25

B

What is the fuel consumption during the cruise leg for Operating Conditions BE-34?

OPCION A: 668 pounds.

OPCION B: 718 pounds.

OPCION C: 737 pounds.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8498 Fig. 21 to 25

C

What is the fuel consumption during the cruise leg for Operating Conditions BE-35?

OPCION A: 900 pounds.

OPCION B: 1,030 pounds.

OPCION C: 954 pounds.

(Ver figura referencial 21 TO 25 en el Manual de Figuras)

8499 Fig. 26

A

What are the time and distance to descend from 18,000 feet to 2,500 feet?

OPCION A: 10.3 minutes, 39 NM.

OPCION B: 9.8 minutes, 33 NM.

OPCION C: 10.0 minutes, 36 NM.

(Ver figura referencial 26 en el Manual de Figuras)

8500 Fig. 26

B

What are the distance and fuel consumption to descend from 22,000 feet to 4,500 feet?

OPCION A: 44 NAM, 117 pounds.

OPCION B: 48 NAM, 112 pounds.

OPCION C: 56 NAM, 125 pounds.

(Ver figura referencial 26 en el Manual de Figuras)

8501 Fig. 26

C

What are the time and distance to descend from 16,500 feet to 3,500 feet?

OPCION A: 9.3 minutes, 37 NAM.

OPCION B: 9.1 minutes, 35 NAM.

OPCION C: 8.7 minutes, 33 NAM.

(Ver figura referencial 26 en el Manual de Figuras)

8502 Fig. 26

B

What are the distance and fuel consumption to descend from 13,500 feet to 1,500 feet?

OPCION A: 30 NAM, 87 pounds.

OPCION B: 29 NAM, 80 pounds.

OPCION C: 38 NAM, 100 pounds.

(Ver figura referencial 26 en el Manual de Figuras)

8503 Fig. 26

C

What are the time and distance to descend from 23,000 feet to 600 feet with an average 15-knot headwind?

OPCION A: 14.2 minutes, 50 NAM.

OPCION B: 14.6 minutes, 56 NAM.

OPCION C: 14.9 minutes, 59 NAM.

(Ver figura referencial 26 en el Manual de Figuras)

8504 Fig. 27 - 28

A

What is the landing distance over a 50-foot obstacle for Operating Conditions B-36?

OPCION A: 1,900 feet.

OPCION B: 1,625 feet.

OPCION C: 950 feet.
(Ver figura referencial 27/28 en el Manual de Figuras)

8505 Fig. 27 - 28 A
What are the approach speed and ground roll when landing under Operating Conditions B-36?

OPCION A: 113 knots and 950 feet.

OPCION B: 113 knots and 1,950 feet.

OPCION C: 112 knots and 900 feet.

(Ver figura referencial 27/28 en el Manual de Figuras)

8506 Fig. 27 - 28 B
What is the remaining runway length when stopped after landing over a 50-foot obstacle for Operating Conditions B-37?

OPCION A: 2,500 feet.

OPCION B: 2,000 feet.

OPCION C: 2,600 feet.

(Ver figura referencial 27/28 en el Manual de Figuras)

8507 Fig. 27 - 28 A
What are the approach speed and ground roll when landing under Operating Conditions B-37?

OPCION A: 108 knots and 1,400 feet.

OPCION B: 109 knots and 900 feet.

OPCION C: 107 knots and 1,350 feet.

(Ver figura referencial 27/28 en el Manual de Figuras)

8508 Fig. 27 - 28 B
What is the landing distance over a 50-foot obstacle for Operating Conditions B-38?

OPCION A: 1,850 feet.

OPCION B: 1,700 feet.

OPCION C: 1,800 feet.

(Ver figura referencial 27/28 en el Manual de Figuras)

8509 Fig. 27 - 28 C
What is the total runway used when touchdown is at the 1,000 foot marker for Operating Conditions B-38?

OPCION A: 2,000 feet.

OPCION B: 1,700 feet.

OPCION C: 1,800 feet.

(Ver figura referencial 27/28 en el Manual de Figuras)

8510 Fig. 27 - 28 C
What is the remaining runway length when stopped after landing over a 50-foot obstacle for Operating Conditions B-39?

OPCION A: 2,300 feet.

OPCION B: 2,400 feet.

OPCION C: 2,500 feet.

(Ver figura referencial 27/28 en el Manual de Figuras)

8511 Fig. 27 - 28 B
What are the approach speed and ground roll when landing under Operating Conditions B-39?

OPCION A: 111 knots and 1,550 feet.

OPCION B: 110 knots and 1,400 feet.

OPCION C: 109 knots and 1,300 feet.

(Ver figura referencial 27/28 en el Manual de Figuras)

8512 Fig. 27 - 28 C
What is the landing distance over a 50-foot obstacle for Operating Conditions B-40?

OPCION A: 1,500 feet.

OPCION B: 1,750 feet.

OPCION C: 1,650 feet.

(Ver figura referencial 27/28 en el Manual de Figuras)

8583 Fig. 45 - 46 - 47

A

What are V1 and Vr speeds for Operating Conditions A-1?

OPCION A: V1 123.1 knots; Vr 125.2 knots.

OPCION B: V1 120.5 knots; Vr 123.5 knots.

OPCION C: V1 122.3 knots; Vr 124.1 knots.

(Ver figura referencial 45/46/47 en el Manual de Figuras)

8584 Fig. 45 - 46 - 47

C

What are V1 and Vr speeds for Operating Conditions A-2?

OPCION A: V1 129.7 knots; Vr 134.0 knots.

OPCION B: V1 127.2 knots; Vr 133.2 knots.

OPCION C: V1 127.4 knots; Vr 133.6 knots.

(Ver figura referencial 45/46/47 en el Manual de Figuras)

8585 Fig. 45 - 46 - 47

A

What are V1 and Vr speeds for Operating Conditions A-3?

OPCION A: V1 136.8 knots; Vr 141.8 knots.

OPCION B: V1 134.8 knots; Vr 139.0 knots.

OPCION C: V1 133.5 knots; Vr 141.0 knots.

(Ver figura referencial 45/46/47 en el Manual de Figuras)

8586 Fig. 45 - 46 - 47

B

What are V1 and Vr speeds for Operating Conditions A-4?

OPCION A: V1 128.0 knots; Vr 130.5 knots.

OPCION B: V1 129.9 knots; Vr 133.4 knots.

OPCION C: V1 128.6 knots; Vr 131.1 knots.

(Ver figura referencial 45/46/47 en el Manual de Figuras)

8587 Fig. 45 - 46 - 47

B

What are V1 and Vr speeds for Operating Conditions A-5?

OPCION A: V1 110.4 knots; Vr 110.9 knots.

OPCION B: V1 109.6 knots; Vr 112.7 knots.

OPCION C: V1 106.4 knots; Vr 106.4 knots.

(Ver figura referencial 45/46/47 en el Manual de Figuras)

8593 Fig. 48 - 49 - 50

A

What is the ground distance covered during en route climb for Operating Conditions W-1?

OPCION A: 104.0 NM.

OPCION B: 99.2 NM.

OPCION C: 109.7 NM.

(Ver figura referencial 48/49/50 en el Manual de Figuras)

8594 Fig. 48 - 49 - 50

C

What is the ground distance covered during en route climb for Operating Conditions W-2?

OPCION A: 85.8 NM.

OPCION B: 87.8 NM.

OPCION C: 79.4 NM.

(Ver figura referencial 48/49/50 en el Manual de Figuras)

8595 Fig. 48 - 49 - 50

A

What is the ground distance covered during en route climb for Operating Conditions W-3?

OPCION A: 86.4 NM.

OPCION B: 84.2 NM.

OPCION C: 85.1 NM.

(Ver figura referencial 48/49/50 en el Manual de Figuras)

8596 Fig. 48 - 49 - 50

B

What is the ground distance covered during en route climb for Operating Conditions W-4?

OPCION A: 58.4 NM.

OPCION B: 61.4 NM.

OPCION C: 60.3 NM.

(Ver figura referencial 48/49/50 en el Manual de Figuras)

8597 Fig. 48 - 49 - 50

C

What is the ground distance covered during en route climb for Operating Conditions W-5?

OPCION A: 68.0 NM.

OPCION B: 73.9 NM.

OPCION C: 66.4 NM.

(Ver figura referencial 48/49/50 en el Manual de Figuras)

8598 Fig. 48 - 49 - 50

B

What is the aircraft weight at the top of climb for Operating Conditions W-1?

OPCION A: 81,600 pounds.

OPCION B: 81,400 pounds.

OPCION C: 81,550 pounds.

(Ver figura referencial 48/49/50 en el Manual de Figuras)

8599 What is the aircraft weight at the top of climb for Operating Conditions W-2? (Refer to Figures 48, 49, and 50)

C

OPCION A: 82,775 pounds.

OPCION B: 83,650 pounds.

OPCION C: 83,800 pounds.

(Ver figura referencial 48,49,50 en el Manual de Figuras)

8600 What is the aircraft weight at the top of climb for Operating Conditions W-3? (Refer to Figures 48, 49, and 50)

B

OPCION A: 75,750 pounds.

OPCION B: 75,900 pounds.

OPCION C: 76,100 pounds.

(Ver figura referencial 48,49,50 en el Manual de Figuras)

8601 What is the aircraft weight at the top of climb for Operating Conditions W-4? (Refer to Figures 48, 49, and 50)

B

OPCION A: 86,150 pounds.

OPCION B: 86,260 pounds.

OPCION C: 86,450 pounds.

(Ver figura referencial 48,49,50 en el Manual de Figuras)

8602 What is the aircraft weight at the top of climb for Operating Conditions W-5? (Refer to Figures 48, 49, and 50)

A

OPCION A: 89,900 pounds.

OPCION B: 90,000 pounds.

OPCION C: 90,100 pounds.

(Ver figura referencial 48,49,50 en el Manual de Figuras)

8603 Fig. 51 - 52

B

What is the total time from starting to the alternate through completing the approach for Operating Conditions L-1?

OPCION A: 30 minutes.

OPCION B: 44 minutes.

OPCION C: 29 minutes.

(Ver figura referencial 51/52 en el Manual de Figuras)

8604 Fig. 51,52

B

What is the total time from starting to the alternate through completing the approach for Operating Conditions L-2?

OPCION A: 36 minutes.

OPCION B: 55 minutes.

OPCION C: 40 minutes.

(Ver figura referencial 51/52 en el Manual de Figuras)

8605 Fig. 51,52

B

What is the total time from starting to the alternate through completing the approach for Operating Conditions L-3?

OPCION A: 1 hour.

OPCION B: 1 hour 15 minutes.

OPCION C: 1 hour 24 minutes.

(Ver figura referencial 51,52 en el Manual de Figuras)

(Ver figura referencial 51/52 en el Manual de Figuras)

8606 Fig. 51,52

A

What is the total time from starting to the alternate through completing the approach for Operating Conditions L-4?

OPCION A: 35 minutes.

OPCION B: 19 minutes.

OPCION C: 20 minutes.

(Ver figura referencial 51/52 en el Manual de Figuras)

8607 Fig. 51,52

A

What is the total time from starting to the alternate through completing the approach for Operating Conditions L-5?

OPCION A: 1 hour 3 minutes.

OPCION B: 48 minutes.

OPCION C: 55 minutes.

(Ver figura referencial 51/52 en el Manual de Figuras)

8608 Fig. 51,52

C

What is the approximate landing weight for Operating Conditions L-1?

OPCION A: 79,000 pounds.

OPCION B: 83,600 pounds.

OPCION C: 81,500 pounds.

(Ver figura referencial 51/52 en el Manual de Figuras)

8609 Fig. 51,52

B

What is the approximate landing weight for Operating Conditions L-2?

OPCION A: 65,200 pounds.

OPCION B: 65,800 pounds.

OPCION C: 69,600 pounds.

(Ver figura referencial 51/52 en el Manual de Figuras)

8610 Fig. 51,52

A

What is the approximate landing weight for Operating Conditions L-3?

OPCION A: 80,300 pounds.

OPCION B: 85,400 pounds.

OPCION C: 77,700 pounds.

(Ver figura referencial 51/52 en el Manual de Figuras)

8611 Fig. 51,52

A

What is the approximate landing weight for Operating Conditions L-4?

OPCION A: 73,200 pounds.

OPCION B: 74,190 pounds.

OPCION C: 73,500 pounds.

(Ver figura referencial 51/52 en el Manual de Figuras)

8612 Fig. 51,52

B

What is the approximate landing weight for Operating Conditions L-5?

OPCION A: 78,600 pounds.

OPCION B: 77,000 pounds.

OPCION C: 76,300 pounds.

(Ver figura referencial 51/52 en el Manual de Figuras)

8613 What is the takeoff EPR for Operating Conditions R-1? (See Figures 53, 54, and 55)

C

OPCION A: 2.04.

OPCION B: 2.01.

OPCION C: 2.035.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8614 What is the takeoff EPR for Operating Conditions R-2? (See Figures 53, 54, and 55)

A

OPCION A: 2.19.

OPCION B: 2.18.

OPCION C: 2.16.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8615 What is the takeoff EPR for Operating Conditions R-3? (See Figures 53, 54, and 55)

C

OPCION A: 2.01.

OPCION B: 2.083.

OPCION C: 2.04.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8616 What is the takeoff EPR for Operating Conditions R-4? (See Figures 53, 54, and 55)

B

OPCION A: 2.06.

OPCION B: 2.105.

OPCION C: 2.11.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8617 What is the takeoff EPR for Operating Conditions R-5? (See Figures 53, 54, and 55)

A

OPCION A: 1.98.

OPCION B: 1.95.

OPCION C: 1.96.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8618 What is the takeoff safety speed for Operating Conditions R-1? (See figures 53, 54, and 55)

A

OPCION A: 128 knots.

OPCION B: 121 knots.

OPCION C: 133 knots.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8619 What is the rotation speed for Operating Conditions R-2? (See figures 53, 54, and 55)

C

OPCION A: 147 knots.

OPCION B: 152 knots.

OPCION C: 146 knots.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8620 What are V1, Vr, and V2 speeds for Operating Conditions R-3? (See figures 53, 54, and 55)

B

OPCION A: 143, 143, and 147 knots.

OPCION B: 138, 138, and 142 knots.

OPCION C: 136, 138, and 143 knots.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8621 What are critical engine failure and takeoff safety speeds for Operating Conditions R-4? (See figures 53, 54, and 55)

B

OPCION A: 131 and 133 knots.

OPCION B: 123 and 134 knots.

OPCION C: 122 and 130 knots.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8622 What are rotation and V2 bug speeds for Operating Conditions R-5? (See figures 53, 54, and 55)

A

OPCION A: 138 and 143 knots.

OPCION B: 136 and 138 knots.

OPCION C: 134 and 141 knots.

(Ver figura referencial 53,54,55 en el Manual de Figuras)

8628 What is the ground distance covered during en route climb for Operating Conditions V-1? (Refer to Figures 56, 57, and 58)

A

OPCION A: 145 NM.

OPCION B: 137 NM.

OPCION C: 134 NM.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8629 What is the ground distance covered during en route climb for Operating Conditions V-2? (Refer to Figures 56, 57, and 58)

C

- OPCION A:** 84 NM.
OPCION B: 65 NM.
OPCION C: 69 NM.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8630 What is the ground distance covered during en route climb for Operating Conditions V-3? (Refer to Figures 56, 57, and 58) B

- OPCION A:** 95 NM.
OPCION B: 79 NM.
OPCION C: 57 NM.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8631 What is the ground distance covered during en route climb for Operating Conditions V-4? (Refer to Figures 56, 57, and 58) A

- OPCION A:** 63 NM.
OPCION B: 53 NM.
OPCION C: 65 NM.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8632 What is the ground distance covered during en route climb for Operating Conditions V-5? (Refer to Figures 56, 57, and 58) C

- OPCION A:** 70 NM.
OPCION B: 47 NM.
OPCION C: 61 NM.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8633 How much fuel is burned during en route climb for Operating Conditions V-1? (Refer to Figures 56, 57, and 58) C

- OPCION A:** 4,100 pounds.
OPCION B: 3,600 pounds.
OPCION C: 4,000 pounds.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8634 How much fuel is burned during en route climb for Operating Conditions V-2? (Refer to Figures 56, 57, and 58) A

- OPCION A:** 2,250 pounds.
OPCION B: 2,600 pounds.
OPCION C: 2,400 pounds.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8635 What is the aircraft weight at the top of climb for Operating Conditions V-3? (Refer to Figures 56, 57, and 58) B

- OPCION A:** 82,100 pounds.
OPCION B: 82,500 pounds.
OPCION C: 82,200 pounds.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8636 What is the aircraft weight at the top of climb for Operating Conditions V-4? (Refer to Figures 56, 57, and 58) A

- OPCION A:** 102,900 pounds.
OPCION B: 102,600 pounds.
OPCION C: 103,100 pounds.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8637 What is the aircraft weight at the top of climb for Operating Conditions V-5? (Refer to Figures 56, 57, and 58) A

- OPCION A:** 73,000 pounds.
OPCION B: 72,900 pounds.
OPCION C: 72,800 pounds.

(Ver figura referencial 56,57,58 en el Manual de Figuras)

8638 What is the maximum climb EPR for Operating Conditions T-1? (See Figures 59 and 60) A

- OPCION A:** 1.82.
OPCION B: 1.96.
OPCION C: 2.04.

(Ver figura referencial 59,60 en el Manual de Figuras)

8639 What is the maximum climb EPR for Operating Conditions T-2? (See Figures 59 and 60)

C

OPCION A: 2.10.

OPCION B: 1.99.

OPCION C: 2.02.

(Ver figura referencial 59,60 en el Manual de Figuras)

8640 What is the maximum climb EPR for Operating Conditions T-3? (See Figures 59 and 60)

C

OPCION A: 2.11.

OPCION B: 2.02.

OPCION C: 1.90.

(Ver figura referencial 59,60 en el Manual de Figuras)

8641 What is the maximum climb EPR for Operating Conditions T-4? (See Figures 59 and 60)

C

OPCION A: 2.20.

OPCION B: 2.07.

OPCION C: 2.06.

(Ver figura referencial 59,60 en el Manual de Figuras)

8642 What is the maximum climb EPR for Operating Conditions T-5? (See Figures 59 and 60)

B

OPCION A: 2.00.

OPCION B: 2.04.

OPCION C: 1.96.

(Ver figura referencial 59,60 en el Manual de Figuras)

8643 Fig. 61,62

C

What is the trip time for Operating Conditions X-1?

OPCION A: 4 hours 5 minutes.

OPCION B: 4 hours 15 minutes.

OPCION C: 4 hours.

(Ver figura referencial 61/62 en el Manual de Figuras)

8644 Fig. 61,62

B

What is the trip time for Operating Conditions X-2?

OPCION A: 5 hours 5 minutes.

OPCION B: 6 hours 15 minutes.

OPCION C: 5 hours 55 minutes.

(Ver figura referencial 61/62 en el Manual de Figuras)

8645 Fig. 61,62

C

What is the trip time for Operating Conditions X-3?

OPCION A: 4 hours 15 minutes.

OPCION B: 3 hours 40 minutes.

OPCION C: 4 hours.

(Ver figura referencial 61/62 en el Manual de Figuras)

8646 Fig. 61,62

B

What is the trip time for Operating Conditions X-4?

OPCION A: 6 hours 50 minutes.

OPCION B: 5 hours 45 minutes.

OPCION C: 5 hours 30 minutes.

(Ver figura referencial 61/62 en el Manual de Figuras)

8647 Fig. 61,62

A

What is the trip time for Operating Conditions X-5?

OPCION A: 2 hours 55 minutes.

OPCION B: 3 hours 10 minutes.

OPCION C: 2 hours 50 minutes.

(Ver figura referencial 61/62 en el Manual de Figuras)

8648 Fig. 61,62

B

What is the trip fuel for Operating Conditions X-1?

OPCION A: 25,000 pounds.

OPCION B: 26,000 pounds.

OPCION C: 24,000 pounds.

(Ver figura referencial 61/62 en el Manual de Figuras)

8649 Fig. 61,62

C

What is the trip fuel for Operating Conditions X-2?

OPCION A: 33,000 pounds.

OPCION B: 28,000 pounds.

OPCION C: 35,000 pounds.

(Ver figura referencial 61/62 en el Manual de Figuras)

8650 Fig. 61,62

B

What is the trip fuel for Operating Conditions X-3?

OPCION A: 36,000 pounds.

OPCION B: 34,500 pounds.

OPCION C: 33,000 pounds.

(Ver figura referencial 61/62 en el Manual de Figuras)

8651 Fig. 61,62

A

What is the trip fuel for Operating Conditions X-4?

OPCION A: 33,000 pounds.

OPCION B: 31,500 pounds.

OPCION C: 34,000 pounds.

(Ver figura referencial 61/62 en el Manual de Figuras)

8652 Fig. 61,62

C

What is the trip fuel for Operating Conditions X-5?

OPCION A: 15,000 pounds.

OPCION B: 20,000 pounds.

OPCION C: 19,000 pounds.

(Ver figura referencial 61/62 en el Manual de Figuras)

8653 Fig. 63,64

C

What is the turbulent air penetration N1 power setting for Operating Conditions Q-1?

OPCION A: 82.4 percent?

OPCION B: 84.0 percent.

OPCION C: 84.8 percent.

(Ver figura referencial 63/64 en el Manual de Figuras)

8654 Fig. 63,64

B

What is the turbulent air penetration N1 power setting for Operating Conditions Q-2?

OPCION A: 78.2 percent.

OPCION B: 75.2 percent.

OPCION C: 76.7 percent.

(Ver figura referencial 63/64 en el Manual de Figuras)

8655 What is the turbulent air penetration N1 power setting for Operating Conditions Q-3?

C

OPCION A: 77.8 percent.

OPCION B: 82.6 percent.

OPCION C: 84.2 percent.

(Ver figura referencial 63/64 en el Manual de Figuras)

8656 Fig. 63,64

A

What is the turbulent air penetration N1 power setting for Operating Conditions Q-4?

OPCION A: 76.8 percent.

OPCION B: 75.4 percent.

OPCION C: 74.0 percent.

(Ver figura referencial 63/64 en el Manual de Figuras)

8658 Fig. 66,67

B

What is the trip time corrected for wind under Operating Conditions Z-1?

OPCION A: 58.1 minutes.

OPCION B: 51.9 minutes.

OPCION C: 54.7 minutes.

(Ver figura referencial 66/67 en el Manual de Figuras)

8659 Fig. 66,67

C

What is the trip time corrected for wind under Operating Conditions Z-2?

OPCION A: 1 hour 35 minutes.

OPCION B: 1 hour 52 minutes.

OPCION C: 1 hour 46 minutes.

(Ver figura referencial 66/67 en el Manual de Figuras)

8661 Fig. 66,67

B

What is the trip time corrected for wind under Operating Conditions Z-4?

OPCION A: 48.3 minutes.

OPCION B: 50.7 minutes.

OPCION C: 51.3 minutes.

(Ver figura referencial 66/67 en el Manual de Figuras)

8662 Fig. 66,67

A

What is the trip time corrected for wind under Operating Conditions Z-5?

OPCION A: 1 hour 11 minutes.

OPCION B: 56 minutes.

OPCION C: 62 minutes.

(Ver figura referencial 66/67 en el Manual de Figuras)

8663 Fig. 66,67

A

What is the estimated fuel consumption for Operating Conditions Z-1?

OPCION A: 5,230 pounds.

OPCION B: 5,970 pounds.

OPCION C: 5,550 pounds.

(Ver figura referencial 66/67 en el Manual de Figuras)

8664 Fig. 66,67

A

What is the estimated fuel consumption for Operating Conditions Z-2?

OPCION A: 10,270 pounds.

OPCION B: 9,660 pounds.

OPCION C: 10,165 pounds.

(Ver figura referencial 66/67 en el Manual de Figuras)

8665 Fig. 66,67

B

What is the estimated fuel consumption for Operating Conditions Z-3?

OPCION A: 12,300 pounds.

OPCION B: 11,300 pounds.

OPCION C: 13,900 pounds.

(Ver figura referencial 66/67 en el Manual de Figuras)

8666 Fig. 66,67

C

What is the estimated fuel consumption for Operating Conditions Z-4?

OPCION A: 4,950 pounds.

OPCION B: 5,380 pounds.

OPCION C: 5,230 pounds.

(Ver figura referencial 66/67 en el Manual de Figuras)

8667 Fig. 66,67

C

What is the estimated fuel consumption for Operating Conditions Z-5?

OPCION A: 6,250 pounds.

OPCION B: 6,250 pounds.

OPCION C: 6,250 pounds.

(Ver figura referencial 66/67 en el Manual de Figuras)

OPCION B: 5,380 pounds.

OPCION C: 7,120 pounds.

(Ver figura referencial 66/67 en el Manual de Figuras)

8668 Fig. 68,69

C

What are the recommended IAS and EPR settings for holding under Operating Conditions O-1?

OPCION A: 221 knots and 1.83 EPR.

OPCION B: 223 knots and 2.01 EPR.

OPCION C: 217 knots and 1.81 EPR.

(Ver figura referencial 68/69 en el Manual de Figuras)

8669 Fig. 68,69

B

What are the recommended IAS and EPR settings for holding under Operating Conditions O-2?

OPCION A: 210 knots and 1.57 EPR.

OPCION B: 210 knots and 1.515 EPR.

OPCION C: 210 knots and 1.45 EPR.

(Ver figura referencial 68/69 en el Manual de Figuras)

8670 Fig. 68,69

B

What are the recommended IAS and EPR settings for holding under Operating Conditions O-3?

OPCION A: 217 knots and 1.50 EPR.

OPCION B: 215 knots and 1.44 EPR.

OPCION C: 216 knots and 1.40 EPR.

(Ver figura referencial 68/69 en el Manual de Figuras)

8671 Fig. 68,69

A

What are the recommended IAS and EPR settings for holding under Operating Conditions O-4?

OPCION A: 223 knots and 1.33 EPR.

OPCION B: 225 knots and 1.33 EPR.

OPCION C: 220 knots and 1.28 EPR.

(Ver figura referencial 68/69 en el Manual de Figuras)

8672 Fig. 68,69

C

What are the recommended IAS and EPR settings for holding under Operating Conditions O-5?

OPCION A: 219 knots and 1.28 EPR.

OPCION B: 214 knots and 1.26 EPR.

OPCION C: 218 knots and 1.27 EPR.

(Ver figura referencial 68/69 en el Manual de Figuras)

8673 Fig. 68,69

A

What is the approximate fuel consumed when holding under Operating Conditions O-1?

OPCION A: 1,625 pounds.

OPCION B: 1,950 pounds.

OPCION C: 2,440 pounds.

(Ver figura referencial 68/69 en el Manual de Figuras)

8674 Fig. 68,69

C

What is the approximate fuel consumed when holding under Operating Conditions O-2?

OPCION A: 2,250 pounds.

OPCION B: 2,500 pounds.

OPCION C: 3,000 pounds.

(Ver figura referencial 68/69 en el Manual de Figuras)

8675 Fig. 68,69

A

What is the approximate fuel consumed when holding under Operating Conditions O-3?

OPCION A: 2,940 pounds.

OPCION B: 2,520 pounds.

OPCION C: 3,250 pounds.

(Ver figura referencial 68/69 en el Manual de Figuras)

8676 Fig.68,69

C

What is the approximate fuel consumed when holding under Operating Conditions O-4?

OPCION A: 2,870 pounds.

OPCION B: 2,230 pounds.

OPCION C: 1,440 pounds.

(Ver figura referencial 68/69 en el Manual de Figuras)

8677 Fig. 68,69

C

What is the approximate fuel consumed when holding under Operating Conditions O-5?

OPCION A: 2,950 pounds.

OPCION B: 2,870 pounds.

OPCION C: 2,400 pounds.

(Ver figura referencial 68/69 en el Manual de Figuras)

8678 Fig. 70

B

How many minutes of dump time is required to reach a weight of 144,500 pounds?

Initial weight 180,500 lb

Zero fuel weight 125,500 lb

OPCION A: 13 minutes.

OPCION B: 15 minutes.

OPCION C: 16 minutes.

(Ver figura referencial 70 en el Manual de Figuras)

8679 Fig. How many minutes of dump time is required to reduce fuel load to 25,000 pounds?

C

Initial weight 179,500 lb

Zero fuel weight 136,500 lb

OPCION A: 10 minutes.

OPCION B: 9 minutes.

OPCION C: 8 minutes.

(Ver figura referencial 70 en el Manual de Figuras)

8680 Fig.70

C

How many minutes of dump time is required to reach a weight of 151,500 pounds?

Initial weight 181,500 lb

Zero fuel weight 126,000 lb

OPCION A: 15 minutes.

OPCION B: 14 minutes.

OPCION C: 13 minutes.

(Ver figura referencial 70 en el Manual de Figuras)

8681 Fig. 70

A

How many minutes of dump time is required to reduce fuel load to 16,000 pounds?

Initial weight 175,500 lb

Zero fuel weight 138,000 lb

OPCION A: 9 minutes.

OPCION B: 10 minutes.

OPCION C: 8 minutes.

(Ver figura referencial 70 en el Manual de Figuras)

8682 Fig. 71,72

A

What is the approximate level-off pressure altitude after drift-down under Operating Conditions D-1?

OPCION A: 19,400 feet.

OPCION B: 18,000 feet.

OPCION C: 20,200 feet.

(Ver figura referencial 71/72 en el Manual de Figuras)

8683 Fig. 71,72

B

What is the approximate level-off pressure altitude after drift-down under Operating Conditions D-2?

OPCION A: 14,700 feet.

OPCION B: 17,500 feet.

OPCION C: 18,300 feet.

(Ver figura referencial 71/72 en el Manual de Figuras)

8684 Fig. 71,72

C

What is the approximate level-off pressure altitude after drift-down under Operating Conditions D-3?

OPCION A: 22,200 feet.

OPCION B: 19,800 feet.

OPCION C: 21,600 feet.

(Ver figura referencial 71/72 en el Manual de Figuras)

8685 Fig. 71,72

C

What is the approximate level-off pressure altitude after drift-down under Operating Conditions D-4?

OPCION A: 27,900 feet.

OPCION B: 22,200 feet.

OPCION C: 24,400 feet.

(Ver figura referencial 71/72 en el Manual de Figuras)

8686 Fig. 71,72

B

What is the approximate level-off pressure altitude after drift-down under Operating Conditions D-5?

OPCION A: 8,800 feet.

OPCION B: 9,600 feet.

OPCION C: 13,000 feet.

(Ver figura referencial 71/72 en el Manual de Figuras)

8687 Fig. 73,75

B

What is the go-around EPR for Operating Conditions L-1?

OPCION A: 2.01 EPR.

OPCION B: 2.03 EPR.

OPCION C: 2.04 EPR.

(Ver figura referencial 73/75 en el Manual de Figuras)

8688 What is the go-around EPR for Operating Conditions L-2?

C

OPCION A: 2.115 EPR.

OPCION B: 2.10 EPR.

OPCION C: 2.06 EPR.

(Ver figura referencial 73/75 en el Manual de Figuras)

8689 What is the go-around EPR for Operating Conditions L-3?

A

OPCION A: 2.06 EPR.

OPCION B: 2.07 EPR.

OPCION C: 2.09 EPR.

(Ver figura referencial 73/75 en el Manual de Figuras)

8690 What is the go-around EPR for Operating Conditions L-4?

A

OPCION A: 2.056 EPR.

OPCION B: 2.12 EPR.

OPCION C: 2.096 EPR.

(Ver figura referencial 73/75 en el Manual de Figuras)

8691 Fig. 73,75

A

What is the go-around EPR for Operating Conditions L-5?

OPCION A: 2.00 EPR.

OPCION B: 2.04 EPR.

OPCION C: 2.05 EPR.

(Ver figura referencial 73/75 en el Manual de Figuras)

8692 Fig. 73,74,75

A

What is Vref for Operating Conditions L-1?

OPCION A: 143 knots.

OPCION B: 144 knots.

OPCION C: 145 knots.

(Ver figura referencial 73/74/75 en el Manual de Figuras)

8693 Fig. 73,74,75

B

What is the reference speed for Operating Conditions L-2?

OPCION A: 140 knots.

OPCION B: 145 knots.

OPCION C: 148 knots.

(Ver figura referencial 73/74/75 en el Manual de Figuras)

8694 Fig. 73,74,75

B

What is Vref +20 for Operating Conditions L-3?

OPCION A: 151 knots.

OPCION B: 169 knots.

OPCION C: 149 knots.

(Ver figura referencial 73/74/75 en el Manual de Figuras)

8695 Fig. 73,74,75

C

What is Vref +10 for Operating Conditions L-4?

OPCION A: 152 knots.

OPCION B: 138 knots.

OPCION C: 148 knots.

(Ver figura referencial 73/74/75 en el Manual de Figuras)

8696 Fig. 73,74,75

C

What is the maneuvering speed for Operating Conditions L-5?

OPCION A: 124 knots.

OPCION B: 137 knots.

OPCION C: 130 knots.

(Ver figura referencial 73/74/75 en el Manual de Figuras)

8712 What is the maximum takeoff EPR for Operating Conditions G-1? (See Figures 81, 82, and 83)

A

OPCION A: Engines 1 and 3, 2.22; engine 2, 2.16.

OPCION B: Engines 1 and 3, 2.22; engine 2, 2.21.

OPCION C: Engines 1 and 3, 2.15; engine 2, 2.09.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8713 What is the maximum takeoff EPR for Operating Conditions G-2? (See Figures 81, 82, and 83)

C

OPCION A: Engines 1 and 3, 2.15; engine 2, 2.16.

OPCION B: Engines 1 and 3, 2.18; engine 2, 2.13.

OPCION C: Engines 1 and 3, 2.14; engine 2, 2.11.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8714 What is the maximum takeoff EPR for Operating Conditions G-3? (See Figures 81, 82, and 83)

B

OPCION A: Engines 1 and 3, 2.08; engine 2, 2.05.

OPCION B: Engines 1 and 3, 2.14; engine 2, 2.10.

OPCION C: Engines 1 and 3, 2.18; engine 2, 2.07.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8715 What is the maximum takeoff EPR for Operating Conditions G-4? (See Figures 81, 82, and 83)

A

OPCION A: Engines 1 and 3, 2.23; engine 2, 2.21.

OPCION B: Engines 1 and 3, 2.26; engine 2, 2.25.

OPCION C: Engines 1 and 3, 2.24; engine 2, 2.24.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8716 What is the maximum takeoff EPR for Operating Conditions G-5? (See Figures 81, 82, and 83)

C

OPCION A: Engines 1 and 3, 2.27; engine 2, 2.18.

OPCION B: Engines 1 and 3, 2.16; engine 2, 2.14.

OPCION C: Engines 1 and 3, 2.23; engine 2, 2.22.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8717 What is the takeoff safety speed for Operating Conditions G-1? (See Figures 81, 82, and 83)

B

OPCION A: 122 knots.

OPCION B: 137 knots.

OPCION C: 133 knots.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8718 What is the rotation speed for Operating Conditions G-2? (See Figures 81, 82, and 83)

C

OPCION A: 150 knots.

OPCION B: 154 knots.

OPCION C: 155 knots.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8719 What are V1, Vr and V2 speeds for Operating Conditions G-3? (See figures 81, 82, and 83)

A

OPCION A: 134, 134, and 145 knots.

OPCION B: 134, 139, and 145 knots.

OPCION C: 132, 132, and 145 knots.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8720 What are V1 and V2 speeds for Operating Conditions G-4? (See figures 81, 82, and 83)

C

OPCION A: 133 and 145 knots.

OPCION B: 127 and 141 knots.

OPCION C: 132 and 146 knots.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8721 What are rotation and V2 bug speeds for Operating Conditions G-5? (See figures 81, 82, and 83)

B

OPCION A: 120 and 134 knots.

OPCION B: 119 and 135 knots.

OPCION C: 135 and 135 knots.

(Ver figura referencial 81,82,83 en el Manual de Figuras)

8727 Fig. 84,85

A

What are the recommended IAS and EPR settings for holding under Operating Conditions H-1?

OPCION A: 264 knots and 1.80 EPR.

OPCION B: 259 knots and 1.73 EPR.

OPCION C: 261 knots and 1.81 EPR.

(Ver figura referencial 84/85 en el Manual de Figuras)

8728 Fig. 84,85

C

What are the recommended IAS and EPR settings for holding under Operating Conditions H-2?

OPCION A: 257 knots and 1.60 EPR.

OPCION B: 258 knots and 1.66 EPR.

OPCION C: 253 knots and 1.57 EPR.

(Ver figura referencial 84/85 en el Manual de Figuras)

8729 Fig. 84,85

B

What are the recommended IAS and EPR settings for holding under Operating Conditions H-3?

OPCION A: 226 knots and 1.30 EPR.

OPCION B: 230 knots and 1.31 EPR.

OPCION C: 234 knots and 1.32 EPR.

(Ver figura referencial 84/85 en el Manual de Figuras)

8730 Fig. 84,85

B

What are the recommended IAS and EPR settings for holding under Operating Conditions H-4?

OPCION A: 219 knots and 1.44 EPR.

OPCION B: 216 knots and 1.42 EPR.

OPCION C: 220 knots and 1.63 EPR.
(Ver figura referencial 84/85 en el Manual de Figuras)

8731 Fig. 84,85

C

What are the recommended IAS and EPR settings for holding under Operating Conditions H-5?

OPCION A: 245 knots andn 1.65 EPR.

OPCION B: 237 knots andn 1.61 EPR.

OPCION C: 249 knots andn 1.67 EPR.

(Ver figura referencial 84/85 en el Manual de Figuras)

8732 Fig. 84,85

C

What is the approximate fuel consumed when holding under Operating Conditions H-1?

OPCION A: 3,500 pounds.

OPCION B: 4,680 pounds.

OPCION C: 2,630 pounds.

(Ver figura referencial 84/85 en el Manual de Figuras)

8733 FIG. 84,85

A

What is the approximate fuel consumed when holding under Operating Conditions H-2?

OPCION A: 5,100 pounds.

OPCION B: 3,400 pounds.

OPCION C: 5,250 pounds.

(Ver figura referencial 84/85 en el Manual de Figuras)

8734 Fig. 84,85

B

What is the approximate fuel consumed when holding under Operating Conditions H-3?

OPCION A: 3,090 pounds.

OPCION B: 6,950 pounds.

OPCION C: 6,680 pounds.

(Ver figura referencial 84/85 en el Manual de Figuras)

8735 Fig. 84,85

A

What is the approximate fuel consumed when holding under Operating Conditions H-4?

OPCION A: 3,190 pounds.

OPCION B: 3,050 pounds.

OPCION C: 2,550 pounds.

(Ver figura referencial 84/85 en el Manual de Figuras)

8736 Fig. 84,85

C

What is the approximate fuel consumed when holding under Operating Conditions H-5?

OPCION A: 3,170 pounds.

OPCION B: 7,380 pounds.

OPCION C: 5,540 pounds.

(Ver figura referencial 84/85 en el Manual de Figuras)

8737 Fig. 86,87

B

What are descent time and distance under Operating Conditions S-1?

OPCION A: 24 minutes, 118 NM.

OPCION B: 26 minutes, 125 NM.

OPCION C: 25 minutes, 118 NM.

(Ver figura referencial 86/87 en el Manual de Figuras)

8738 Fig. 86,87

B

What are descent fuel and distance under Operating Conditions S-2?

OPCION A: 1,440 pounds, 104 NM.

OPCION B: 1,500 pounds, 118 NM.

OPCION C: 1,400 pounds, 98 NM.

(Ver figura referencial 86/87 en el Manual de Figuras)

8739 Fig. 86,87

A

What are descent fuel and distance under Operating Conditions S-3?

OPCION A: 1,490 pounds, 118 NM.

OPCION B: 1,440 pounds, 110 NM.

OPCION C: 1,550 pounds, 127 NM.

(Ver figura referencial 86/87 en el Manual de Figuras)

8740 Fig. 86,87

B

What are descent time and distance under Operating Conditions S-4?

OPCION A: 22 minutes, 110 NM.

OPCION B: 21 minutes, 113 NM.

OPCION C: 24 minutes, 129 NM.

(Ver figura referencial 86/87 en el Manual de Figuras)

8741 Fig. 86,87

A

What are descent fuel and distance under Operating Conditions S-5?

OPCION A: 1,420 pounds, 97 NAM.

OPCION B: 1,440 pounds, 102 NAM.

OPCION C: 1,390 pounds, 92 NAM.

(Ver figura referencial 86/87 en el Manual de Figuras)

8742 Fig. 88,89

B

Which conditions will result in the shortest landing distance at a weight of 132,500 pounds?

OPCION A: Dry runway using brakes and reversers.

OPCION B: Dry runway using brakes and spoilers.

OPCION C: Wet runway using brakes, spoilers and reversers.

(Ver figura referencial 88/89 en el Manual de Figuras)

8743 Fig. 88

C

How much longer is the dry runway landing distance using brakes only compared to using brakes and reversers at 114,000 pounds gross weight?

OPCION A: 1,150 feet.

OPCION B: 500 feet.

OPCION C: 300 feet.

(Ver figura referencial 88 en el Manual de Figuras)

8744 Figure 88

B

How many feet will remain after landing on a 7,200-foot dry runway with spoilers inoperative at 118,000 pounds gross weight?

OPCION A: 4,200 feet.

OPCION B: 4,500 feet.

OPCION C: 4,750 feet.

(Ver figura referencial 88 en el Manual de Figuras)

8745 Figure 88

B

What is the maximum landing weight which will permit stopping 2,000 feet short of the end of a 5,400-foot dry runway with reversers and spoilers inoperative?

OPCION A: 117,500 pounds.

OPCION B: 136,500 pounds.

OPCION C: 139,500 pounds.

(Ver figura referencial 88 en el Manual de Figuras)

8746 Fig. 89

C

Which of the following configurations will result in the shortest landing distance over a 50-foot obstacle to a wet runway?

OPCION A: Brakes and spoilers at 122,500 pounds gross weight.

OPCION B: Brakes and reversers at 124,000 pounds gross weight.

OPCION C: Brakes, spoilers, and reversers at 131,000 pounds gross weight.

(Ver figura referencial 89 en el Manual de Figuras)

8747 Fig. 89 B
How many feet will remain after landing on a 6,000-foot wet runway with reversers inoperative at 122,000 pounds gross weight?

OPCION A: 2,200 feet.

OPCION B: 2,750 feet.

OPCION C: 3,150 feet.

(Ver figura referencial 89 en el Manual de Figuras)

8748 Fig. 90 C
Which configuration will result in a landing distance of 5,900 feet over a 50 foot obstacle to an icy runway?

OPCION A: Use of three reversers at 131,000 pounds gross weight.

OPCION B: Use of brakes and spoilers at 125,000 pounds gross weight.

OPCION C: Use of three reversers at 133,000 pounds gross weight.

(Ver figura referencial 90 en el Manual de Figuras)

8749 Fig. 90 A
What is the transition distance when landing on an icy runway at a gross weight of 134,000 pounds?

OPCION A: 400 feet.

OPCION B: 950 feet.

OPCION C: 1,350 feet.

(Ver figura referencial 90 en el Manual de Figuras)

8750 Fig. 90 B
What is the maximum landing weight which will permit stopping 700 feet short of the end of a 5,200-foot icy runway?

OPCION A: 124,000 pounds.

OPCION B: 137,000 pounds.

OPCION C: 108,000 pounds.

(Ver figura referencial 90 en el Manual de Figuras)

8751 Fig. 90 C
What is the landing distance on an icy runway with reversers inoperative at a landing weight of 125,000 pounds?

OPCION A: 4,500 feet.

OPCION B: 4,750 feet.

OPCION C: 5,800 feet.

(Ver figura referencial 90 en el Manual de Figuras)

8752 Fig. 91 B
How much will landing distance be reduced by using 15° of flaps rather than 0° flaps at a landing weight of 119,000 pounds?

OPCION A: 500 feet.

OPCION B: 800 feet.

OPCION C: 2,700 feet.

(Ver figura referencial 91 en el Manual de Figuras)

8753 Fig. 91 A
What is the ground roll when landing with 15° of flaps at a landing weight of 122,000 pounds?

OPCION A: 1,750 feet.

OPCION B: 2,200 feet.

OPCION C: 2,750 feet.

(Ver figura referencial 91 en el Manual de Figuras)

8754 Fig. 91,92 C
What approach speed and ground roll will be needed when landing at a weight of 140,000 pounds if flaps are not used?

OPCION A: 138 knots and 3,900 feet.

OPCION B: 153 knots and 2,900 feet.

OPCION C: 183 knots and 2,900 feet.

(Ver figura referencial 91/92 en el Manual de Figuras)

8755 Fig. 91 A
How much more runway will be used to land with 0° flaps rather than 15° of flaps at a landing weight of 126,000 pounds?

- OPCION A:** 900 feet.
OPCION B: 1,800 feet.
OPCION C: 2,700 feet.

(Ver figura referencial 91 en el Manual de Figuras)

8756 Fig. 91,92 B
What approach speed and landing distance will be needed when landing at a weight of 140,000 pounds with 15° of flaps?

- OPCION A:** 123 knots and 3,050 feet.
OPCION B: 138 knots and 3,050 feet.
OPCION C: 153 knots and 2,050 feet.

(Ver figura referencial 91/92 en el Manual de Figuras)

8757 Fig. 92 C
What is the maximum charted indicated airspeed while maintaining a 3° glide slope at a weight of 140,000 pounds?

- OPCION A:** 127 knots.
OPCION B: 149 knots.
OPCION C: 156 knots.

(Ver figura referencial 92 en el Manual de Figuras)

8758 Fig.92 B
What is the thrust required to maintain a 3° glide slope at 140,000 pounds, with gear down, flaps 30°, and an airspeed of $V_{ref} + 30$ knots?

- OPCION A:** 13,300 pounds.
OPCION B: 16,200 pounds.
OPCION C: 17,700 pounds.

(Ver figura referencial 92 en el Manual de Figuras)

8759 Fig. 92 B
What thrust is required to maintain level flight at 140,000 pounds, with gear up, flaps 25°, and an airspeed of 172 knots?

- OPCION A:** 13,700 pounds.
OPCION B: 18,600 pounds.
OPCION C: 22,000 pounds.

(Ver figura referencial 92 en el Manual de Figuras)

8760 Fig. 92 B
What thrust is required to maintain level flight at 140,000 pounds, with gear down, flaps 25°, and an airspeed of 162 knots?

- OPCION A:** 17,400 pounds.
OPCION B: 19,500 pounds.
OPCION C: 22,200 pounds.

(Ver figura referencial 92 en el Manual de Figuras)

8761 Fig. 92 B
What thrust is required to maintain level flight at 140,000 pounds, with gear down, flaps 25°, and an airspeed of 145 knots?

- OPCION A:** 16,500 pounds.
OPCION B: 18,100 pounds.
OPCION C: 18,500 pounds.

(Ver figura referencial 92 en el Manual de Figuras)

8762 Fig. 92 A
What is the change of total drag for a 140,000-pound airplane when configuration is changed from flaps 30°, gear down, to flaps 0°, gear up, at a constant airspeed of 160 knots?

- OPCION A:** 13,500 pounds.
OPCION B: 13,300 pounds.
-

OPCION C: 15,300 pounds.
(Ver figura referencial 92 en el Manual de Figuras)

8763 Fig. 93

What is the maximum charted indicated airspeed while maintaining a 3° glide slope at a weight of 110,000 pounds?

A

OPCION A: 136 knots.

OPCION B: 132 knots.

OPCION C: 139 knots.

(Ver figura referencial 93 en el Manual de Figuras)

8764 Fig. 93

What is the thrust required to maintain a 3° glide slope at 110,000 pounds, with gear down, flaps 30°, and an airspeed of $V_{ref} + 20$ knots?

B

OPCION A: 9,800 pounds.

OPCION B: 11,200 pounds.

OPCION C: 17,000 pounds.

(Ver figura referencial 93 en el Manual de Figuras)

8765 Fig. 93

What thrust is required to maintain level flight at 110,000 pounds, with gear down, flaps 40°, and an airspeed of 118 knots?

B

OPCION A: 17,000 pounds.

OPCION B: 20,800 pounds.

OPCION C: 22,300 pounds.

(Ver figura referencial 93 en el Manual de Figuras)

8766 Fig. 93

What thrust is required to maintain level flight at 110,000 pounds, with gear up, flaps 25°, and an airspeed of 152 knots?

A

OPCION A: 14,500 pounds.

OPCION B: 15,900 pounds.

OPCION C: 16,700 pounds.

(Ver figura referencial 93 en el Manual de Figuras)

8774 The maximum speed during takeoff that the pilot may abort the takeoff and stop the airplane within the accelerate-stop distance is

C

OPCION A: V_2 .

OPCION B: V_{ef} .

OPCION C: V_1 .

8775 The minimum speed during takeoff, following a failure of the critical engine at V_{ef} , at which the pilot may continue the takeoff and achieve the required height above the takeoff surface within the takeoff distance is indicated by symbol

B

OPCION A: V_{2min} .

OPCION B: V_1 .

OPCION C: V_{lof} .

8780 The symbol for the speed at which the critical engine is assumed to fail during takeoff is

C

OPCION A: V_2 .

OPCION B: V_1 .

OPCION C: V_{ef} .

8795 Fig. 103

What CAS should be used to maintain the fixed TAS at the proposed altitude?

B

OPCION A: 157 knots.

OPCION B: 167 knots.

OPCION C: 172 knots.

(Ver figura referencial 103 en el Manual de Figuras)

8835 (Refer to Figures 115, 116, and 117.)

B

Due to traffic, LAX Center radar vectored PTL 130 to TRM, then cleared the flight to PHX via J169 BLH, direct to Arlin Intersection. What approximate indicated MACH should be maintained to arrive over the BLH VORTAC 8 minutes after passing TRM VORTAC?

OPCION A: .84 Mach.

OPCION B: .82 Mach.

OPCION C: .86 Mach.

(Ver figura referencial 115,116 en el Manual de Figuras)

8933 A definition of the term "viscous hydroplaning" is where

B

OPCION A: the airplane rides on standing water.

OPCION B: a film of moisture covers the painted or rubber-coated portion of the runway.

OPCION C: the tires of the airplane are actually riding on a mixture of steam and melted rubber.

8934 Which term describes the hydroplaning which occurs when an airplane's tire is effectively held off a smooth runway surface by steam generated by friction?

A

OPCION A: Reverted rubber hydroplaning.

OPCION B: Dynamic hydroplaning.

OPCION C: Viscous hydroplaning.

8935 At what minimum speed (rounded off) could dynamic hydroplaning occur on main tires having a pressure of 121 psi?

B

OPCION A: 90 knots.

OPCION B: 96 knots.

OPCION C: 110 knots.

8936 At what minimum speed will dynamic hydroplaning begin if a tire has an air pressure of 70 psi?

C

OPCION A: 85 knots.

OPCION B: 80 knots.

OPCION C: 75 knots.

8937 What is the best method of speed reduction if hydroplaning is experienced on landing?

C

OPCION A: Apply full main wheel braking only.

OPCION B: Apply nosewheel and main wheel braking alternately and abruptly.

OPCION C: Apply aerodynamic braking to the fullest advantage.

8938 Compared to dynamic hydroplaning, at what speed does viscous hydroplaning occur when landing on a smooth, wet runway?

B

OPCION A: At approximately 2.0 times the speed that dynamic hydroplanning occurs.

OPCION B: At a lower speed than dynamic hydroplaning.

OPCION C: At the same speed as dynamic hydroplaning.

8939 What effect, if any, will landing at a higher-than-recommended touchdown speed have on hydroplaning?

C

OPCION A: No effect on hydroplaning, but increases landing roll.

OPCION B: Reduces hydroplaning potential if heavy braking is applied.

OPCION C: Increases hydroplaning potential regardless of braking.

9058 Which place in the turbojet engine is subjected to the highest temperature?

C

OPCION A: Compressor discharge.

OPCION B: Fuel spray nozzles.

OPCION C: Turbine inlet.

9059 What effect would a change in ambient temperature of air density have on gas-turbine-engine performance?

C

OPCION A: As air density decreases, thrust increases.

OPCION B: As temperature increases, thrust increases.

OPCION C: As temperature increases, thrust decreases.

9060 The most important restriction to the operation of turbojet or turboprop engines is

B

OPCION A: limiting compressor speed.

OPCION B: limiting exhaust gas temperature.	
OPCION C: limiting torque.	

9061 An outside air pressure decreases, thrust output will	C
OPCION A: increase due to greater efficiency of jet aircraft in thin air.	
OPCION B: remain the same since compression of inlet air will compensate for any decrease in air pressure.	
OPCION C: decrease due to higher density altitude.	

9062 What effect will an increase in altitude have upon the available equivalent shaft horsepower (ESHP) of a turboprop engine?	A
OPCION A: Lower air density and engine mass flow will cause a decrease in power.	
OPCION B: Higher propeller efficiency will cause an increase in usable power (ESHP) and thrust.	
OPCION C: Power will remain the same but propeller efficiency will decrease.	

9063 What effect, if any, does high ambient temperature have upon the thrust output of a turbine engine?	A
OPCION A: Thrust will be reduced due to the decrease in air density.	
OPCION B: Thrust will remain the same, but turbine temperature will be higher.	
OPCION C: Thrust will be higher because more heat energy is extracted from the hotter air.	

9064 What characterizes a transient compressor stall?	C
OPCION A: Loud, steady roar accompanied by heavy shuddering.	
OPCION B: Sudden loss of thrust accompanied by a loud whine.	
OPCION C: Intermittent "bang", as backfires and flow reversals take place.	

9065 What indicates that a compressor stall has developed and become steady?	A
OPCION A: Strong vibrations and loud roar.	
OPCION B: Occasional loud "bang" and low reversal.	
OPCION C: Completes loss of power with severe reduction in airspeed.	

9066 Which type of compressor stall has the greatest potential for severe engine damage?	C
OPCION A: Intermittent "backfire" stall.	
OPCION B: Transient "backfire" stall.	
OPCION C: Steady, continuous flow reversal stall.	

9067 What recovery would be appropriate in the event of compressor stall?	A
OPCION A: Reduce fuel flow, reduce angle of attack, and increase airspeed.	
OPCION B: Advance throttle, lower angle of attack, and reduce airspeed.	
OPCION C: Reduce throttle, reduce airspeed, and increase angle of attack.	

9068 Under normal operating conditions, which combination of MAP and RPM produce the most severe wear, fatigue, and damage to high performance reciprocating engines?	A
OPCION A: High RPM and low MAP.	
OPCION B: Low RPM and high MAP.	
OPCION C: High RPM and high MAP.	

9069 What effect does high relative humidity have upon the maximum power output of modern aircraft engines?	B
OPCION A: Neither turbojet nor reciprocating engines are affected.	
OPCION B: Reciprocating engines will experience a significant loss of BHP.	
OPCION C: Turbojet engines will experience a significant loss of thrust.	

9070 Equivalent shaft horsepower (ESHP) of a turbo-prop engine is a measure of	B
OPCION A: turbine inlet temperature.	
OPCION B: shaft horsepower and jet thrust.	
OPCION C: propeller thrust only.	

9071 Minimum specific fuel consumption of the turbo-prop engine is normally available in which altitude range?	B
OPCION A: 10,000 feet to 25,000 feet.	
OPCION B: 25,000 feet to the tropopause.	
OPCION C: The tropopause to 45,000 feet.	

9072	Where is the critical altitude of a supercharged reciprocating engine?	A
OPCION A:	The highest altitude at which a desired manifold pressure can be obtained.	
OPCION B:	Highest altitude where the mixture can be leaned to best power ratio.	
OPCION C:	The altitude at which maximum allowable BMEP can be obtained.	
<hr/>		
9073	What is controlled by the waste gas of a turbo-charged reciprocating engine?	B
OPCION A:	Supercharger gear ratio.	
OPCION B:	Exhaust gas discharge.	
OPCION C:	Throttle opening.	
<hr/>		
9074	How should thrust reversers be applied to reduce landing distance for turbojet aircraft?	A
OPCION A:	Immediately after ground contact.	
OPCION B:	Immediately prior to touchdown.	
OPCION C:	After applying maximum wheel braking.	
<hr/>		
9075	Which condition reduces the required runway for takeoff?	C
OPCION A:	Higher-than-recommended airspeed before rotation.	
OPCION B:	Lower-than-standard air density.	
OPCION C:	Increased headwind component.	
<hr/>		
9076	Which performance factor decreases as airplane gross weight increases, for a given runway?	A
OPCION A:	Critical engine failure speed.	
OPCION B:	Rotation speed.	
OPCION C:	Accelerate-stop distance.	
<hr/>		
9077	Maximum range performance of a turbojet aircraft is obtained by which procedure as aircraft weight reduces?	B
OPCION A:	Increasing speed or altitude.	
OPCION B:	Increasing altitude or decreasing speed.	
OPCION C:	Increasing speed or decreasing altitude.	
<hr/>		
9078	Which procedure produces the minimum fuel consumption for a given leg of the cruise flight?	A
OPCION A:	Increase speed for a headwind.	
OPCION B:	Increase speed for a tailwind.	
OPCION C:	Increase altitude for a headwind, decrease altitude for a tailwind.	
<hr/>		
9079	How should reverse thrust propellers be used during landing for maximum effectiveness in stopping?	B
OPCION A:	Gradually increase reverse power to maximum as rollout speed decreases.	
OPCION B:	Use maximum reverse power as soon as possible after touchdown.	
OPCION C:	Select reverse-pitch after landing and use idle power setting of the engines.	
<hr/>		
9083	What effect does an uphill runway slope have upon takeoff performance?	A
OPCION A:	Increases takeoff distance.	
OPCION B:	Decreases takeoff speed.	
OPCION C:	Decreases takeoff distance.	
<hr/>		
9084	Under which condition during the landing roll are the main wheel brakes at maximum effectiveness?	A
OPCION A:	When wing lift has been reduced.	
OPCION B:	At high groundspeeds.	
OPCION C:	When the wheels are locked and skidding.	
<hr/>		
9085	Which condition has the effect of reducing critical engine failure speed?	A
OPCION A:	Slush on the runway or inoperative antiskid.	
OPCION B:	Low gross weight.	
OPCION C:	High density altitude.	
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9128	What action is appropriate when encountering the first ripple of reported clear air turbulence (CAT)?	C
OPCION A:	Extend flaps to decrease wing loading.	
OPCION B:	Extend gear to provide more drag and increase stability.	
OPCION C:	Adjust airspeed to that recommended for rough air.	

9129	If severe turbulence is encountered, which procedure is recommended?	B
OPCION A:	Maintain a constant altitude.	
OPCION B:	Maintain a constant attitude.	
OPCION C:	Maintain constant airspeed and altitude.	

9317	Which is the definition of V2 speed?	B
OPCION A:	Takeoff decision speed.	
OPCION B:	Takeoff safety speed.	
OPCION C:	Minimum takeoff speed.	

9319	What is the correct symbol for minimum unstick speed?	A
OPCION A:	Vmu.	
OPCION B:	Vmd.	
OPCION C:	Vfc.	

9321	Which is the correct symbol for design cruising speed?	A
OPCION A:	Vc.	
OPCION B:	Vs.	
OPCION C:	Vma.	

9322	Which is the correct symbol for the minimum steady-flight speed or stalling speed in the landing configuration?	C
OPCION A:	Vs.	
OPCION B:	Vs1.	
OPCION C:	Vso.	

9323	Which is correct symbol for the stalling speed or the minimum steady flight speed at which the airplane is controllable?	B
OPCION A:	Vso.	
OPCION B:	Vs.	
OPCION C:	Vs1.	

9324	What is the name of a plane beyond the end of a runway which does not contain obstructions and can be considered when calculating takeoff performance of turbine-powered aircraft?	A
OPCION A:	Clearway.	
OPCION B:	Stopway.	
OPCION C:	Obstruction clearance plane.	

9327	What is an area identified by the term "stopway"?	B
OPCION A:	An area, at least the same width as the runway, capable of supporting an airplane during a normal takeoff.	
OPCION B:	An area designated for use in decelerating an aborted takeoff.	
OPCION C:	An area, not as wide as the runway, capable of supporting an airplane during a normal takeoff.	

9355	Which operational requirement must be observed by a commercial operator when ferrying a large, three-engine, turbojet-powered airplane from one facility to another to repair an inoperative engine?	C
OPCION A:	The computed takeoff distance to reach V1 must not exceed 70 percent of the effective runway length.	
OPCION B:	The existing and forecast weather for departure, en route, and approach must be VFR.	
OPCION C:	No passengers may be carried.	

9358	A commercial operator plans to ferry a large, four-engine, reciprocating-engine-powered airplane from one facility to another to repair an inoperative engine. Which is an operational requirement for the three-engine flight?	B
OPCION A:	The gross weight at takeoff may not exceed 75 percent of the maximum certificated gross weight.	
OPCION B:	Weather conditions at the takeoff and destination airports must be VFR.	
OPCION C:	The computed takeoff distance to reach V1 must not exceed 70 percent of the effective runway length.	

9359	Which operational requirement must be observed when ferrying an air carrier airplane when one of its three turbine engines is inoperative?	A
OPCION A:	The weather conditions at takeoff and destination must be VFR.	
OPCION B:	The flight cannot be conducted between official sunset and official sunrise.	
OPCION C:	Weather conditions must exceed the basic VFR minimums for the entire route, including takeoff and landing.	

9360 Which operational requirement must be observed when ferrying a large, turbine-engine-powered airplane when one of its engines is inoperative? A

OPCION A: The weather conditions at takeoff and destination must be VFR.

OPCION B: Weather conditions must exceed the basic VFR minimums for the entire route, including takeoff and landing.

OPCION C: The flight cannot be conducted between official sunset and sunrise.

9361 When a turbine-engine-powered airplane is to be ferried to another base for repair of an inoperative engine, which operational requirement must be observed? A

OPCION A: Only the required flight crewmembers may be on board the airplane.

OPCION B: The existing and forecast weather for departure, en route, and approach must be VFR.

OPCION C: No passengers except authorized maintenance personnel may be carried.

9546 Fig.94,95,96 A
What is the ETE from Chicago Midway Airport to Greater Buffalo Int'l?

OPCION A: 2 hours 12 minutes.

OPCION B: 2 hours 15 minutes.

OPCION C: 2 hours 18 minutes.

(Ver figura referencial 94/95/96 en el Manual de Figuras)

9547 Fig. 94,95,96 A
What are the fuel requirements from Chicago Midway Airport to the Greater Buffalo Int'l?

OPCION A: 2,224 pounds.

OPCION B: 1,987 pounds.

OPCION C: 1,454 pounds.

(Ver figura referencial 94/95/96 en el Manual de Figuras)

9548 Fig. 94,95,96 C
What TAS should be maintained to arrive over CRL VORTAC 42 minutes after level-off?

OPCION A: 166 knots.

OPCION B: 168 knots.

OPCION C: 171 knots.

(Ver figura referencial 94/95/96 en el Manual de Figuras)

9556 Fig.98,100,102 B
What is the ETE from DFW Int'l to IAH?

OPCION A: 1 hour 2 minutes.

OPCION B: 1 hour 4 minutes.

OPCION C: 1 hour 6 minutes.

(Ver figura referencial 98-100/102 en el Manual de Figuras)

9557 Fig.98-100/102 B
What is the total fuel required from DFW Int'l to IAH?

OPCION A: 1,555 pounds.

OPCION B: 1,863 pounds.

OPCION C: 1,941 pounds.

(Ver figura referencial 98-100/102 en el Manual de Figuras)

9558 Fig.98-100/102 B
Determine the TAS required to arrive at CUGAR, 31 minutes after level-off?

OPCION A: 269 knots.

OPCION B: 264 knots.

OPCION C: 258 knots.

(Ver figura referencial 98-100/102 en el Manual de Figuras)

9559 Fig. 98,100,102 A
Determine the TAS required to arrive at CUGAR, 29 minutes after level-off?

OPCION A: 285 knots.

OPCION B: 290 knots.

OPCION C: 295 knots.

(Ver figura referencial 98-100/102 en el Manual de Figuras)

9560 Fig. 103,104,105,106

B

Estimate the total fuel required to be on the aircraft, prior to taxi at Tucson Int'l.

OPCION A: 2,223 pounds.

OPCION B: 2,327 pounds.

OPCION C: 2,447 pounds.

(Ver figura referencial 103 TO 106 en el Manual de Figuras)

9561 Fig. 103,104,105,106

B

Determine the ETE for the flight from Tucson Int'l to Los Angeles Int'l.

OPCION A: 2 hours 10 minutes.

OPCION B: 2 hours 15 minutes.

OPCION C: 2 hours 19 minutes.

(Ver figura referencial 103 TO 106 en el Manual de Figuras)

9579 (Refer to Figures 115, 116, 117, 118, and 118C)

B

What is the total fuel required at .78 Mach?

OPCION A: 22,140 pounds.

OPCION B: 22,556 pounds.

OPCION C: 22,972 pounds.

9580 (Refer to Figures 115, 116, 117, 118, and 118C)

B

What is the specific range in nautical miles per 1,000 pounds of fuel from level-off to the ARLIN Intersection using .78 Mach?

OPCION A: 46.1 NAM/1,000 pounds.

OPCION B: 48.2 NAM/1,000 pounds.

OPCION C: 50.0 NAM/1,000 pounds.

9581 (Refer to Figures 115, 116, 117, 118, and 118C)

C

What is the ETE at .80 Mach?

OPCION A: 1 hour 02 minutes.

OPCION B: 1 hour 04 minutes.

OPCION C: 1 hour 07 minutes.

9582 (Refer to Figures 115, 116, 117, 118, and 118C)

B

What is the total fuel required at .80 Mach?

OPCION A: 22,836 pounds.

OPCION B: 22,420 pounds.

OPCION C: 22,256 pounds.

9583 (Refer to Figures 115, 116, 117, 118, and 118C)

C

What approximate indicated Mach should be maintained to arrive over the BZA VORTAC 6 minutes after passing IPL VORTAC?

OPCION A: .73 Mach.

OPCION B: .74 Mach.

OPCION C: .715 Mach.

9584 (Refer to Figures 107, 115, 116, 117, 118, and 118C)

B

What is the ETE at .82 Mach?

OPCION A: 1 hour 05 minutes.

OPCION B: 1 hour 07 minutes.

OPCION C: 1 hour 03 minutes.



9585 (Refer to Figures 115, 116, 117, 118, and 118C)

B

What is the total fuel required at .82 Mach?

- OPCION A:** 22,420 pounds.
OPCION B: 22,284 pounds.
OPCION C: 22,700 pounds.

9591 Fig. 119 to 122

C

What is the ETE from BUF to ORD using .78 Mach?

- OPCION A:** 1 hour 09 minutes.
OPCION B: 1 hour 07 minutes.
OPCION C: 1 hour 05 minutes.

(Ver figura referencial 119 TO 122 en el Manual de Figuras)

9592 Fig. 119 to 122

B

What is the total fuel required for the flight from BUF to ORD using .78 Mach?

- OPCION A:** 19,033 pounds.
OPCION B: 21,739 pounds.
OPCION C: 22,189 pounds.

(Ver figura referencial 119 TO 122 en el Manual de Figuras)

9593 Fig. 119 to 122

A

What is the specific range in nautical miles per 1,000 pounds of fuel from level-off to start of descent using .78 Mach?

- OPCION A:** 48.8 NAM/1000.
OPCION B: 52.5 NAM/1000.
OPCION C: 55.9 NAM/1000.

(Ver figura referencial 119 TO 122 en el Manual de Figuras)

9594 Fig. 119 to 122

B

What is the ETE from BUF to ORD using .80 Mach?

- OPCION A:** 1 hour 01 minutes.
OPCION B: 1 hour 04 minutes.
OPCION C: 1 hour 08 minutes.

(Ver figura referencial 119 TO 122 en el Manual de Figuras)

9595 Fig. 119 to 122

C

What is the total fuel required for the flight from BUF to ORD using .80 Mach?

- OPCION A:** 19,388 pounds.
OPCION B: 22,094 pounds.
OPCION C: 21,644 pounds.

(Ver figura referencial 119 TO 122 en el Manual de Figuras)

9598 (Refer to Figures 158, 159, 160, 160A, and 161)

B

The required amount of fuel (in pounds) to be on N711JB, prior to taxi, is

- OPCION A:** 5,993 pounds.
OPCION B: 6,408 pounds.
OPCION C: 6,641 pounds.
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